

SUPPLEMENTAL TYPE CERTIFICATE

10039017 REV. 2

This Supplemental Type Certificate is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EU) No. 748/2012 to

BERINGER AERO S.A.S.

AEROPOLE 05130 TALLARD FRANCE

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified below:

Original Type Certificate Number: EASA.A.304

Type Certificate Holder: SN CENTRAIR

Type: 101 "PEGASE"

Model: 101 A, 101, 101 AP, 101 B

101 BC, 101 D, 101 P

Description of Design Change:

Installation of main wheels and brake system P/N: JA-01.

Revision 1: Update with new wheel P/N: RF-002(B) with brake EA-006E supersedes the P/N: JA-01.

Revision 2: Update with new wheel P/N: RF-019 supersedes the wheel P/N: RF-002(B).

EASA Certification Basis:

The Certification Basis (CB) for the original product remains applicable to this certificate/approval.

The requirements for environmental protection and the associated certified noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval.

See Continuation Sheet(s)

For the European Aviation Safety Agency

Cologne, Germany, 21 September 2017

Dominique ROLAND

Head of General Aviation and

Remotely Piloted Aircraft Systems (RPAS)





Associated Technical Documentation:

- Modification Dossier: DM-STC-003 Rev 2 dated 29/06/2017.

- Maintenance checks: MC-STC-003 Rev 2 dated 29/06/2017.

Limitations/Conditions:

Prior to installation of this design change it must be determined that the interrelationship between this design change and any other previously installed design change and/or repair will introduce no adverse effect upon the airworthiness of the product.

- End -